



Revista Facultad de Ingeniería Universidad de Antioquia

ISSN: 0120-6230

ISSN: 2422-2844

Facultad de Ingeniería, Universidad de Antioquia

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Revista Facultad de Ingeniería Universidad de
Antioquia, no. 104, 2022, July-September, pp. 101-107
Facultad de Ingeniería, Universidad de Antioquia

DOI: <https://doi.org/10.17533/udea.redin.20210739>

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Three-Bar structure optimality criterion using the linear resizing rule

Criterio de optimalidad en estructuras de tres barras empleando la regla lineal de redimensionado

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CITE THIS ARTICLE AS:

J. A. Ramírez-Monares and J. I. Hernández-Hernández.

"Three-Bar structure optimality criterion using the linear resizing rule", *Revista Facultad de Ingeniería Universidad de Antioquia*, no. 104, pp. 101-107, Jul-Sep 2022. [Online]. Available: <https://www.doi.org/10.17533/udea.redin.20210850>

ARTICLE INFO:

Received: September 17, 2020

Accepted: August 31, 2021

Available online: August 31, 2021

KEYWORDS:

Optimization; structural design; algorithms; equations

optimización; diseño estructural; algoritmos; ecuaciones

ABSTRACT: The static analysis of the indeterminate three-bar structure is developed using the Castigliano's first theorem, taking the lengths and inclination angles as variables. Some reductions are applied in the resulting set of Equations to approximate them to the references models. From now on, the minimum mass optimization model with restrictions is established. Then, the Optimality Criterion linear resizing optimization rule algorithm for the unbounded and bounded design variables is applied in two numerical cases. The analytical and Matlab Optimization Toolbox results are also obtained and they demonstrate the Optimality Criterion linear resizing rule effectiveness in structural optimization with a minimum mass objective and size restrictions.

RESUMEN: El análisis estático de la estructura indeterminada de tres barras es desarrollada usando el primer teorema de Castigliano, tomando las longitudes y ángulos de inclinación como variables. Al conjunto de ecuaciones obtenidas se aplican algunas reducciones para asemejarlo a los modelos de las referencias. En lo sucesivo, se establece el modelo de optimización de mínima masa con restricciones. Entonces, se aplica el algoritmo de optimización con redimensionamiento lineal del criterio de optimalidad para las variables de diseño restringidas y no restringidas en dos casos numéricos. Los resultados analíticos y de la Matlab Optimization Toolbox también son obtenidos y éstos demuestran la efectividad de la regla de redimensionamiento lineal del criterio de optimalidad en optimización estructural con objetivo de mínima masa y restricciones de dimensiones.

1. Introduction

The Optimality Criteria (OC) methods trace their origin to intuitive traditional approaches to the problem of strength design, more directly to the Fully Stress Design (FSD) criteria, with its associated stress ratio resizing algorithm. The FSD criterion states that a structure is of minimum weight if every member is at its maximum allowable

stresses, or at its minimum size at least under one of the loading conditions. This criterion is based on the intuitive, yet incomplete assumption that if a structure is sized so as not to fail either locally or globally under its critical loads, then that structure must be optimum [1]. Later on, the procedures characterized as mathematics programming formulations appeared on the scene. To define them in a succinct way, they look for the minimum or maximum, depending on the objective function, of a multi-variable function subject to limitations (restrictions) expressed by equalities or inequalities. The representation of inequality constraints is important as it allows for identifying the design as that in which not all

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ISSN 0120-6230

e-ISSN 2422-2844

the elements are subject to limited conditions under a load system. This avoids a limitation that is inherent in some of the above methods. The idea of characterizing an optimum structure through conditions that are believed to exist at an optimum and then applying a resizing procedure that directly satisfies those conditions is the fundamental approach normally used by those designers which became formalized as the Optimality Criterion methods, which are classified as parametric structural optimization. Such Optimality criterion consists of discretizing a pre-established structure to find the optimal dimensions of the structure. The design variables that can be modified are the cross-sectional area of each element, lengths, thicknesses and notch radii, unlike topological optimization where holes or cavities are introduced in the structure that were not present at the beginning.

In [2] and [3] it is used an optimality criteria method that proposed a uniform distribution of strain energy in optimal trusses. In [4] it is derived an optimality criterion based on the Kuhn-Tucker conditions for problems with a single restriction, yet unfortunately, it was inapplicable to problems with multiple restrictions. In the 1970-1980 decades, several authors propose various algorithms to identify active restrictions based on the estimation of Lagrange multipliers of the Kuhn-Tucker conditions in problems with multiple restrictions. A unifying vision of all of them is offered in [5], [6] and [7]. However, the problems inherent in the active restrictions identifying methods were not solved rigorously until the development of the dual formulation, in which the identification of active constraints was substantial and intimately linked to the algorithm of mathematical programming used to solve the problem of minimization. From this moment on, the dual formulation was the unifying link between the Optimality Criterion and the mathematical programming techniques. It should be noted that so far, algorithms based on dual formulation, such as the OC does, have only been able to test its effectiveness in structural optimization problems where the optimum is conditioned by a relatively low number of restrictions [6].

There are several works related to optimality criteria in structural design; for example, the work by [8] where the optimality criterion is applied to the design of lattice structures, the work by [9] using a different resizing rule on the design of laminated composites. In the present work, the optimality criterion is used in the design of truss structures implementing the linear resizing rules with $n_g = 1$ and $n_g = 2$ constraints. They can be implemented with more constraints, taking into account the loss of effectiveness of the method in the face of a high number of restrictions. The constraints formulation is obtained from the statics of the structure, in this case, an indeterminate structure, in which Equations of static

equilibrium are not sufficient to determine all the forces and reactions. The resizing rules presented in this work are based on the assumption that the load distribution in the structure is independent of the member sizes and that the loads carried by the members remain constant. So, the linear resizing rule is used to determine the optimal design variables in such a way that minimize the objective function satisfying the restrictions.

2. Description of the mechanical model

Figure 1 shows the mechanical system analyzed herein. The sectional areas and normal stresses in the lateral bars between nodes $A-D$, $B-D$ and $C-D$ are A_{AD} , A_{BD} , A_{CD} and σ_{AD} , σ_{BD} , σ_{CD} , respectively. The $A-D$ and $C-D$ lateral bars show rotations θ_1 and θ_2 with respect to the vertical $B-D$. The $B-D$ bar features length L , and the lateral bars feature lengths $L_{AD} = L / \cos(\theta_1)$, $L_{CD} = L / \cos(\theta_2)$. All the bars are built from the same linearly elastic material featuring modulus of elasticity E . It is clear that the truss has only two degrees of freedom for joint translation, namely the D_1 and D_2 horizontal and vertical translations at joint D , as shown in Figures 2 and 3.

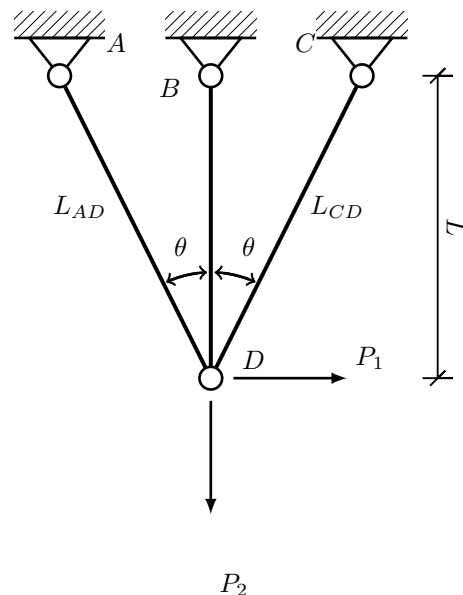
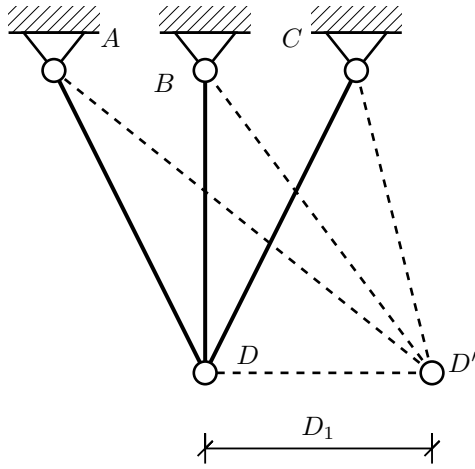
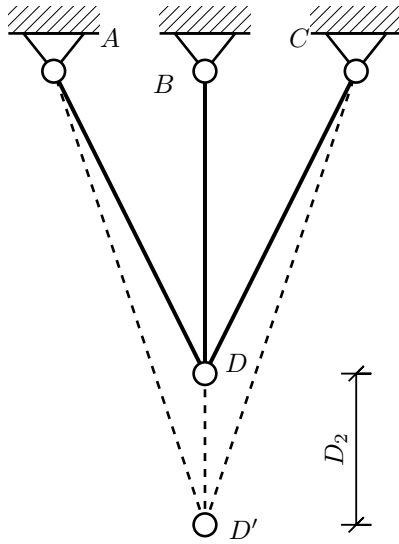


Figure 1 Three bars structure

By applying Castigliano's first theorem to the strain energies of the displacements D_1 and D_2 [10], we obtain the following Equations 1 and 2 for the joint displacements:

$$D_1 = \frac{D_{11}}{D^*} \quad (1)$$

Figure 2 Translations due to P_1 Figure 3 Translations due to P_2

$$D_2 = \frac{D_{22}}{D^*} \quad (2)$$

where D_{11} , D_{22} and D^* are presented in equations 3, 4 and 5

$$D_{11} = -2 \left\{ A_{BD} L_{AD} L_{CD} P_1 + L_{BD} [A_{AD} L_{CD} P_1 \cos^2(\theta_1) - A_{AD} A_{BD} L_{CD} \cos(2\theta_1) - A_{CD} \{ A_{BD} L_{AD} \cos(2\theta_2) - A_{AD} L_{CD} P_2 \cos(\theta_1) \sin(\theta_1) + A_{CD} L_{AD} \cos(\theta_2) (P_1 \cos(\theta_2) + P_2 \sin(\theta_2)) \}] \right\} \quad (3)$$

$$D_{22} = -2 L_{BD} \left\{ -A_{AD} L_{CD} P_1 \cos(\theta_1) \sin(\theta_1) + A_{AD} L_{CD} P_2 \sin^2(\theta_1) + A_{CD} L_{AD} \sin(\theta_2) [P_1 \cos(\theta_2) + P_2 \sin(\theta_2)] \right\} \quad (4)$$

$$D^* = E \left\{ -A_{CD} (A_{BD} L_{AD} + A_{AD} L_{BD}) - A_{AD} A_{BD} L_{CD} + A_{AD} A_{BD} L_{CD} \cos(2\theta_1) + A_{BD} A_{CD} L_{AD} \cos(2\theta_2) + A_{AD} A_{CD} L_{BD} \cos[2(\theta_1 + \theta_2)] + A_{AD} A_{CD} L_{BD} \cos[2(\theta_1 + \theta_2)] \right\} \quad (5)$$

and the stresses in the bars in Equations 6, 7 and 8

$$\sigma_{AD} = \frac{\sigma_{AD1}}{\sigma^*} \quad (6)$$

$$\sigma_{BD} = \frac{\sigma_{BD1}}{\sigma^*} \quad (7)$$

$$\sigma_{CD} = \frac{\sigma_{CD1}}{\sigma^*} \quad (8)$$

where σ_{AD1} , σ_{BD1} , σ_{CD1} and σ^* are shown in Equations 9, 10, 11 and 12.

$$\sigma_{AD1} = A_{CD} L_{BD} P_2 \cos(\theta_1) - A_{CD} L_{BD} P_2 \cos(\theta_1 + 2\theta_2) + P_1 [A_{CD} L_{BD} \sin(\theta_1) + 2A_{BD} L_{CD} \sin(\theta_1) + A_{CD} L_{BD} \sin(\theta_1 + 2\theta_2)] \quad (9)$$

$$\sigma_{BD1} = -2 \left\{ -A_{AD} L_{CD} P_1 \cos(\theta_1) \sin(\theta_1) + A_{AD} L_{CD} P_2 \sin^2(\theta_1) + A_{CD} L_{AD} \sin(\theta_2) [P_1 \cos(\theta_2) + P_2 \sin(\theta_2)] \right\} \quad (10)$$

$$\sigma_{CD1} = 2A_{BD} L_{AD} P_1 \sin(\theta_2) + A_{AD} L_{BD} [P_1 \cos(\theta_1) - P_2 \sin(\theta_1)] \sin(\theta_1 + \theta_2) \quad (11)$$

$$\sigma^* = A_{BD} A_{CD} L_{AD} + A_{AD} A_{CD} L_{BD} + A_{AD} A_{BD} L_{CD} - A_{AD} A_{BD} L_{CD} \cos(2\theta_1) - A_{CD} \{ A_{BD} L_{AD} \cos(2\theta_2) + A_{AD} L_{BD} \cos[2(\theta_1 + \theta_2)] \} \quad (12)$$

2.1 Definition of the optimization model

The combination of the P_1 and P_2 forces can represent the components of any force acting on node D ; however, only positive forces are considered here according to the directions described in Figure 1. The presence of the positive P_1 and P_2 forces implies that the normal stress in the $A-D$ bar is always positive, but the stress in the

$C-D$ bar can be positive or negative. For the sake of model reduction, some symmetries are introduced: the sectional areas and rotation angles of the lateral bars of the $A-D$ and $C-D$ nodes are equal; they are $A_{AD} = A_{CD}$ and $\theta_{AD} = \theta_{CD}$ respectively. The new design variables are $x_1 = A_{AD}0.002E$, $x_2 = A_{BD}0.002E$, $\theta = \theta_{AD} = \theta_{CD}$. The optimization problem consists of finding a minimum-mass design for the truss, given by 13

$$W = \rho L \left[x_2 + \frac{2x_1}{\cos(\theta)} \right] \quad (13)$$

where ρ is the mass density. The truss is designed in subjection to 5 constraints shown in Equations 14 to 18; g_1 corresponds to the vertical displacement at node D and can not exceed a value of $D_{2max} = 0.001L$. The g_2 to g_4 constraints are related to limits on the traction stresses $\sigma_{max}^t = 0.002E$ and finally g_5 is related to compression stress limit in the bar $C-D$ $\sigma_{max}^c = 0.0015E$

$$g_1 = 1 - \frac{D_1}{0.001L} \geq 0 \quad (14)$$

$$g_2 = 1 - \frac{\sigma_{AD}}{0.002E} \geq 0 \quad (15)$$

$$g_3 = 1 - \frac{\sigma_{BD}}{0.002E} \geq 0 \quad (16)$$

$$g_4 = 1 - \frac{\sigma_{CD}}{0.002E} \geq 0 \quad (17)$$

$$g_5 = 1 + \frac{\sigma_{CD}}{0.0015E} \geq 0 \quad (18)$$

Considering the previous reductions, the design problem may be rewritten as Equations 19 - 24

$$\text{Min } \frac{2x_1}{\cos \theta} + x_2 \quad \text{s. t.} \quad (19)$$

$$g_1(x) = 1 - \frac{0.002LP_2}{[x_2 + 2x_1 \cos^3(\theta)]} \geq 0 \quad (20)$$

$$g_2(x) = 1 - \frac{g_{11}}{x_1 g^*} \geq 0 \quad (21)$$

$$g_3(x) = 1 - \frac{g_{22}}{g^*} \geq 0 \quad (22)$$

$$g_4(x) = 1 + \frac{0.004g_{33}}{x_1 g^*} \geq 0 \quad (23)$$

$$g_5(x) = 1 - \frac{0.003g_{33}}{x_1 g^*} \geq 0 \quad (24)$$

where g_{11} , g_{22} , g_{33} and g^* are presented in Equations 25-28

$$g_{11} = 0.002E \left\{ x_1 P_2 \cos(\theta) - x_1 P_2 \cos(3\theta) + P_1 [x_1 + 2x_2 \sec(\theta)] \sin(\theta) + x_1 P_1 \sin(3\theta) \right\} \quad (25)$$

$$g_{22} = 0.002E \{ 4P_2 \sec(\theta) \sin^2(\theta) \} \quad (26)$$

$$g_{33} = E \left\{ x_2 P_1 \sec(\theta) \sin(\theta) + x_1 [P_1 \cos(\theta) - P_2 \sin(\theta)] \sin(2\theta) \right\} \quad (27)$$

$$g^* = x_1 - x_1 \cos(4\theta) + 4x_2 \sec(\theta) \sin^2(\theta) \quad (28)$$

3. Optimization algorithm

The optimization problem can be represented by 29

$$\begin{aligned} &\text{Min } f(y) \\ &\text{subject to } g_j(y) \geq 0, \quad j = 1, 2, \dots, n_g \end{aligned} \quad (29)$$

In this work the objective function is linear. For these problems the Kuhn-Tucker condition is 30

$$\frac{\partial f}{\partial y_i} - \sum_{j=1}^{n_g} \lambda_j \frac{\partial g_j}{\partial y_i} = 0 \quad i = 1, 2, \dots, n \quad (30)$$

In terms of the original design variables, Equation 30 becomes 31

$$x_k^2 f_k - \sum_{j=1}^{n_g} c_{kj} \lambda_j = 0 \quad k = 1, 2, \dots, n \quad (31)$$

where f_k and c_{kj} are presented in 32

$$f_k = \frac{\partial f}{\partial x_k}, \quad c_{kj} = -\frac{\partial g_j}{\partial y_k}, \quad k = 1, 2, \dots, n \quad (32)$$

3.1 Resizing approximation rules

Another possibility for rewriting eq 31 is the exponential rule 33

$$x_k^{new} = x_k \left(\frac{1}{x_k^2 f_k} \sum_{j=1}^{n_g} \lambda_j c_{kj} \right)^{1/\eta} \quad k = 1, 2, \dots, n. \quad (33)$$

where the old value for x_k is used to produce a new estimate. A linearized form of eq 33 obtained by binomial expansion is presented in 34

$$x_k^{new} = x_k + \Delta x_k, \quad k = 1, 2, \dots, n. \quad (34)$$

where Δx_k and λ_j are described in 35 and 36

$$\Delta x_k = \frac{1}{\eta} \left(\frac{1}{x_k^2 f_k} \sum_{j=1}^{n_g} \lambda_j c_{kj} - 1 \right) x_k \quad k = 1, 2, \dots, n. \quad (35)$$

$$\sum_{j=1}^{n_g} \sum_{k=1}^n \frac{c_{kl} c_{kj}}{x_k^3 f_k} \lambda_j = \sum_{k=1}^n \frac{c_{kl}}{x_k} - \eta g_l(x) \quad l = 1, 2, \dots, n_g. \quad (36)$$

the term η is a step size parameter. This linearized form is known as the linear resizing rule.

3.2 Upper and lower limits on design variables

In many cases, it is necessary to have lower and upper limits on design variables besides the displacement constraints. The set of design variables that are at their lower or upper limits during iterations is called the passive set, I_p , while the set including the rest of the variables is called the active set, I_a . Such design variables call for some modifications on the resizing algorithm. Thus, the Lagrange multiplier λ is shown in Equation 37

$$\lambda = \left[\frac{1}{c_0^*} \sum_{i \in I_a} \left(-\frac{\partial f}{\partial x_i} \frac{\partial g}{\partial y_i} \right)^{1/2} \right]^2 \quad (37)$$

where c_0^* and c_0 are presented in 38 and 39

$$c_0^* = c_0 + \sum_{i \in I_p} \frac{\partial g}{\partial y_i} \frac{1}{x_i} \quad (38)$$

$$c_0 = g(x_0) + \sum_{i=1}^n x_{0i} \frac{\partial g}{\partial x_i} \quad (39)$$

and the resizing Equation 40 is obtained from Equation 31

$$x_i = \left(-\lambda \frac{\partial g / \partial y_i}{\partial f / \partial x_i} \right)^{1/2} \quad i = 1, 2, \dots, n. \quad (40)$$

4. Results

This section shows the results obtained by the linear resizing rule with and without limits on the design variables. Two different cases are considered depending on the θ angle, the P_1 , P_2 loads and the objective function in order to adapt the models to those presented in [6] and [7]. The first example, from [6], shows the results of a three-bar truss with the objective function 41

$$W = 4x_1 + x_2 \quad (41)$$

and two restrictions: the vertical displacement and the stress in the $A-D$ bar; these are Equations 19 and 21. This example also considers $P_2 = 8P_1$, $\theta = \pi/3$ and $x_0 = (1, 10)^T$ as a starting point. The reference shows the results only for the first three iterations. They are shown in Table 1 below, where the maximal difference between [6] and the results obtained here is 14.7% at the first

iteration when estimating x_1 . Figures 4 and 5 show the plotted values for the first 10 iterations. It widely shows the fast convergence to the limit values of $x_1 = 0.6598$, $x_2 = 15.835$ and a cost function $f = 18.474358$.

Table 1 Comparison of the first iteration values

iter.	var.	Pres. Met.	[6]	Difference (%)
2	x_1	0.4845	0.5557	14.706
	x_2	13.8125	13.906	0.677
	f	15.7503	16.128	2.398
3	x_1	0.6132	0.6434	4.922
	x_2	15.5798	15.6	0.1295
	f	18.0326	18.1736	0.7814

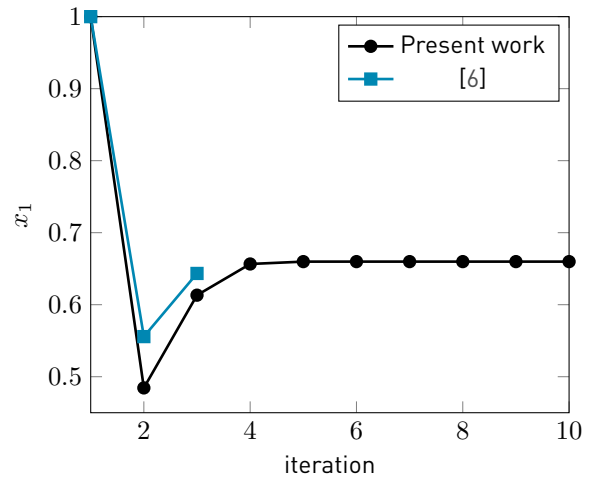


Figure 4 Convergence of the algorithm for x_1 .

The second three-bar truss optimization structure is presented by [7]. This optimization problem considers a $D_{2max} = 0.001$ inch displacement restriction on the vertical direction, with equal forces $P_1 = P_2 = 300/\sqrt{2}$ lb in the vertical and horizontal directions, $\theta = \pi/4$ angle of rotation of the lateral bars, $\rho = 0.283$ lb/in³ density, $E = 30 \times 10^6$ psi elasticity modulus, $L = 100\sqrt{2}$ inch length and $x_0 = (2, 2)^T$ as starting point. So, the design variables are $x_1 = A_{AD}E/P_1l = A_{CD}E/P_1l$ and $x_2 = A_{BD}E/P_1l$.

Two lower limits in the design variables $x_1 \geq 0.1$ and $x_2 \geq 0.1$ are considered in [7]. The present work considers only the $x_1 \geq 0.1$ limit as a passive restriction. The results for the first seven iterations are shown in Table 2. Figure 6 plots the first 7 x_2 iteration values presented by [6] with a blue squared line, and the values here obtained with a black dotted line.

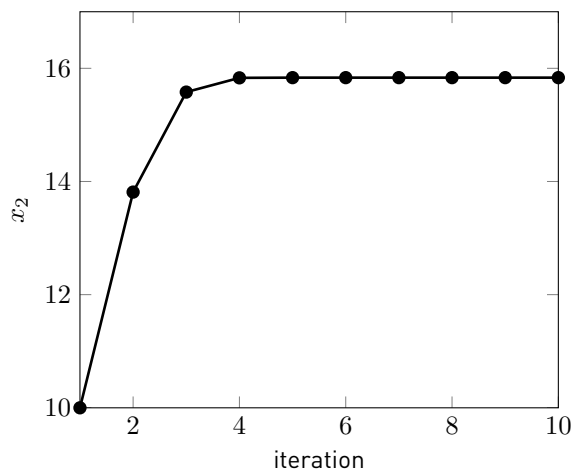


Figure 5 Convergence of the algorithm for x_2 .

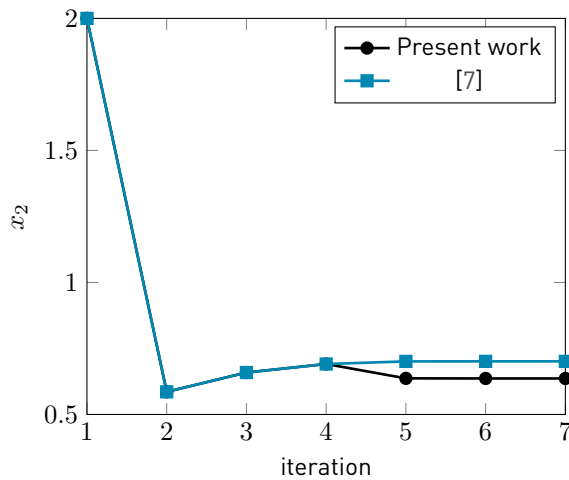


Figure 6 Convergence of the algorithm for x_2 .

5. Conclusions

The results of the first example shown in Figures 4 and 5 were computed in 0.125364 seconds. The Matlab Optimization Toolbox was used to solve this example as well, and, after 34 iterations in 9.2498 seconds, the results were $x_1 = 0.6598288$, $x_2 = 15.852457$ and $f = 18.491773$. The analytical solution of this example can be found by equalling both constraints to zero and solving the system for the variables; when doing so, these results are $x_1 = 0.659726$, $x_2 = 15.8525$ and $f = 18.4914$. This example displayed a fast convergence of the reciprocal exponential resizing rule, as expected, because there are only two restrictions, and both of them are of the passive type.

Regarding the second example, Figure 6 shows a separation between the plot lines from the 5th iteration; such difference is around 0.101%. The difference in the

Table 2 Comparison of the first 7 iteration values

iter.	var.	Pres. Met.	[7]	Difference (%)
2	x_1	0.29289	0.29289	0.00109578
	x_2	0.58578	0.58579	0.00060360
	f	40.022247	40.022090	0.00039186
3	x_1	0.16471	0.16472	0.00109578
	x_2	0.65886	0.65886	0.00041057
	f	31.83039	31.830668	0.00085317
4	x_1	0.1	0.1	0
	x_2	0.69114	0.69115	0.00000596
	f	27.563993	27.563995	0.00000423
5	x_1	0.1	0.1	0
	x_2	0.63679	0.70102	9.16154206
	f	26.025769	27.843316	6.52776434
6	x_1	0.1	0.1	0
	x_2	0.63639	0.70117	9.23796966
	f	26.01446	27.847561	6.58262522
7	x_1	0.1	0.1	0
	x_2	0.63639	0.70117	9.23797323
	f	26.01445	27.847561	6.58262776

values of the cost function is remarkable. The results for the 7th iteration by [7] are $f = 27.847560$; for the values here presented, they are $f = 26.014459$, the difference being 6.58%. Therefore, it can be said that this work proposes a better cost function with less computation effort. Because the results were known a priori, a conclusion was reached that the optimal solution has the tendency to take the lowest limit in x_1 , but not in x_2 , only the x_1 boundary was included in this work's optimization model. The Matlab Optimization Toolbox was used to solve this example as well, and after 27 iterations in 9.4198 seconds, the results were $f = 26.014458$, $x_1 = 0.1$ and $x_2 = 0.636396$.

6. Declaration of competing interest

We declare that we have no significant competing interests, including financial or non-financial, professional, or personal interests interfering with the full and objective presentation of the work described in this manuscript.

7. Acknowledgements

We thank to the PRODEP scholarship and to the SNI, the institutions that supported our research activities during the development of this article.

8. Funding

The author(s) received no financial support for the research, authorship, and/or publication of this article.

9. Author contributions

José Alfredo Ramírez Monares: The static analysis of the presented structures and the implementation of the optimality criterion linear resizing rule in the mechanical systems optimization scheme herein.

Jesús Israel Hernández Hernández: The algorithm implementation of the of the optimality criterion linear resizing rule in a computer system.

10. Data availability statement

The authors confirm that the data supporting the findings of this study are available within the article [and/or] its supplementary materials.

References

- [1] L. Berke and N. Khot, "Structural optimization using optimality criteria," in *Computer aided optimal design: structural and mechanical systems*. Springer, 1987, pp. 271–311.
- [2] V. Venkayya, N. Khot, and V. Reddy, "Energy distribution in an optimum structural design," AIR FORCE FLIGHT DYNAMICS LAB WRIGHT-PATTERSON AFB OHIO, Tech. Rep., 1969.
- [3] V. Venkayya, N. Khot, V. Tischer, and R. Taylor, "Design of optimum structures for dynamic loads," in *Proceedings of 3rd. Conf. Matrix Meth. Struct. Mech., Ohio*, 1971.
- [4] L. Berke, "An efficient approach to the minimum weight design of deflection limited structures," *Rep. Air Force Flight Dynamics Lab*, 1970.
- [5] N. Khot, *Optimality criterion methods in structural optimization*. Flight Dynamics Laboratory, Air Force Wright Aeronautical Laboratories, Air Force Systems Command, 1982, vol. 81, no. 3124.
- [6] R. T. Haftka and Z. Gürdal, *Elements of structural optimization*. Springer Science & Business Media, 2012, vol. 11.
- [7] S. S. Rao, *Engineering optimization: theory and practice*. John Wiley & Sons, 2019.
- [8] T. Stanković, J. Mueller, P. Egan, and K. Shea, "A generalized optimality criteria method for optimization of additively manufactured multimaterial lattice structures," *Journal of Mechanical Design*, vol. 137, no. 11, 2015.
- [9] R. Kussmaul, M. Zogg, and P. Ermanni, "An optimality criteria-based algorithm for efficient design optimization of laminated composites using concurrent resizing and scaling," *Structural and Multidisciplinary Optimization*, vol. 58, no. 2, pp. 735–750, 2018.
- [10] S. P. Timoshenko and J. M. Gere, *Mechanics of Materials*, 3rd ed. Van Nostrand Company, 1972.